

the theory of catastrophes. Change in X in the process of control ($u(t) = 0$) and disturbances ($z(t) = 0$) to zero, i.e. free movement of the system can be written in the following form:

$$\frac{dX}{dt} = f(X) \quad (5)$$

Here $f(\bullet)$ - a vector-function operating in the space, which defines a vector $X = (X_1, \dots, X_n)^T$. This vector-function is usually nonlinear.

Derived above standard state X_s is a particular solution of equation (5). Consequently:

$$\frac{dX_s}{dt} = f(X_s)$$

Using this representation from (5) we can obtain equation for x :

$$\frac{dx}{dt} = f(X_s + x) - f(X_s)$$

Naturally, this equation can be expanded, i.e. its right-hand side near any standard state X_s . If a f has the form of a polynomial in terms of x , then it is always possible that leads to a finite number of terms. However, in more complex cases f may depend on X in any other way. In this case, we assume that:

$$\begin{aligned} \frac{dx_i}{dt} = & \sum_{j=1}^n \left. \frac{\partial f_i(X)}{\partial X_j} \right|_{X_s} x_j + \frac{1}{2} \sum_{j=1}^n \sum_{k=1}^n \left. \frac{\partial^2 f_i(X)}{\partial X_j \partial X_k} \right|_{X_s} x_j x_k + \frac{1}{6} \sum_{j=1}^n \sum_{k=1}^n \sum_{m=1}^n \left. \frac{\partial^3 f_i(X)}{\partial X_j \partial X_k \partial X_m} \right|_{X_s} \cdot \\ & \cdot x_j x_k x_m + \frac{1}{24} \sum_{j=1}^n \sum_{k=1}^n \sum_{m=1}^n \sum_{l=1}^n \left. \frac{\partial^4 f_i(X)}{\partial X_j \partial X_k \partial X_m \partial X_l} \right|_{X_s} x_j x_k x_m x_l + \dots \quad i = 1, \dots, n \end{aligned}$$

(7)

Nonlinear stationary control object with m inputs and n outputs, using a smooth change of variables in the canonical form can be written as:

$$\begin{aligned} \frac{dx_i}{dt} = & \left. \frac{\partial f_i(X)}{\partial X_1} \right|_{X_s} x_1 + \left. \frac{\partial f_i(X)}{\partial X_2} \right|_{X_s} x_2 + \dots - x_i^3 - x_{i+1}^3 - \left. \frac{\partial^2 f_i(X)}{\partial X_i \partial X_{i+1}} \right|_{X_s} x_i x_{i+1} + \left. \frac{\partial f_i(X)}{\partial X_{i+1}} \right|_{X_s} = a_{i,i+1}, \dots, \left. \frac{\partial f_i(X)}{\partial X_{n-1}} \right|_{X_s} = a_{i,n-1}, \left. \frac{\partial f_i(X)}{\partial X_n} \right|_{X_s} = a_{i,n}, \\ & + \left. \frac{\partial f_i(X)}{\partial X_{i+1}} \right|_{X_s} x_{i+1}, \dots, \left. \frac{\partial f_i(X)}{\partial X_{n-1}} \right|_{X_s} x_{n-1} + \left. \frac{\partial f_i(X)}{\partial X_n} \right|_{X_s} x_n + \dots \quad i = 1, \dots, n \end{aligned}$$

(8)

where,

1) f continues to be expanded in a power series of X and 2) the expansion can be truncated at power of a finite order. By virtue of the latter assumption, usually we have to be sometimes limited to the study of infinitesimal stability, i.e. study of the system response to small perturbations, such that $|x|/|X_s| \ll 1$. This limitation is often of minor importance, since the infinitesimal stability gives a necessary condition for instability in the sense that if X_s is unstable with respect to small x , then it will be unstable with respect to any x .

Formally, the above-described expansion in state space $x(t) \in R^n$ can be represented by the equation:

$$\frac{dx}{dt} = \left(\frac{\partial f}{\partial X} \right)_{x_s} x + \frac{1}{2} \left(\frac{\partial^2 f}{\partial X \partial X} \right)_{x_s} xx + \frac{1}{6} \left(\frac{\partial^3 f}{\partial X \partial X \partial X} \right)_{x_s} xxx + \dots \quad (6)$$

Equations (6) corresponds to the description of the free movement of system, comprising deviations x relative to the standard state X_s .

The state equations of the control object in the expanded form can be represented in the following form:

$$\left. \frac{\partial f_i(X)}{\partial X_1} \right|_{X_s} = a_{i1}, \left. \frac{\partial f_i(X)}{\partial X_2} \right|_{X_s} = a_{i2}, \left. \frac{\partial f_i(X)}{\partial X_i} \right|_{X_s} = a_{ii},$$

$$\left. \frac{\partial f_i(X)}{\partial X_{i+1}} \right|_{X_s} = a_{i,i+1}, \dots, \left. \frac{\partial f_i(X)}{\partial X_{n-1}} \right|_{X_s} = a_{i,n-1}, \left. \frac{\partial f_i(X)}{\partial X_n} \right|_{X_s} = a_{in},$$

$$\left. \frac{\partial f_i(X)}{\partial X_i \partial X_{i+1}} \right|_{X_s} = k_{i,k+1}^o$$

Nonlinear stationary control system (1) subject to the control procedure (2) and a mathematical model of control object (8) in the expanded form can be presented as the following system of equations:

$$\left\{ \begin{aligned} \dot{x}_1 &= -b_{11}x_1^3 - b_{11}x_2^3 - b_{11}k_{12}x_1x_2 + (a_{11} + b_{11}k_1)x_1 + (a_{12} + b_{11}k_2)x_2 + a_{13}x_3 + \dots + a_{1n}x_n \\ \dot{x}_2 &= -b_{22}x_1^3 - b_{22}x_2^3 - b_{22}k_{12}x_1x_2 + (a_{21} + b_{22}k_1)x_1 + (a_{22} + b_{22}k_2)x_2 + a_{23}x_3 + \dots + a_{2n}x_n \\ \dot{x}_3 &= -b_{33}x_3^3 - b_{33}x_4^3 - b_{33}k_{34}x_3x_4 + a_{32}x_2 + (a_{33} + b_{33}k_3)x_3 + (a_{34} + b_{33}k_4)x_4 + a_{35}x_5 + \dots + a_{3n}x_n \\ \dot{x}_4 &= -b_{44}x_3^3 - b_{44}x_4^3 - b_{44}k_{34}x_3x_4 + a_{41}x_1 + a_{42}x_2 + (a_{43} + b_{44}k_3)x_3 + (a_{44} + b_{44}k_4)x_4 + \\ &+ a_{45}x_5 + \dots + a_{4n}x_n \\ &\dots \dots \dots \dots \dots \dots \dots \\ \dot{x}_{n-1} &= -b_{n-1,n-1}x_{n-1}^3 - b_{n-1,n-1}x_n^3 - b_{n-1,n-1}k_{n-1,n}x_{n-1}x_n + a_{n1}x_1 + a_{n2}x_2 + \dots + (a_{n-1,n-1} + b_{n-1,n-1}k_{n-1})x_{n-1} + \\ &+ (a_{n-1,n} + b_{n-1,n}k_n)x_n \\ \dot{x}_n &= -b_{nn}x_n^3 - b_{nn}x_{n-1}^3 - b_{nn}k_{n-1,n}x_{n-1}x_n + a_{n1}x_1 + a_{n2}x_2 + \dots + (a_{n,n-1} + b_{nn}k_{n-1})x_{n-1} + (a_{n,n} + b_{nn}k_n)x_n \end{aligned} \right. \quad (9)$$

where $k_{i,i+1}^o + k_{i,i+1}^c = k_{i,i+1}$, $i = 1, \dots, n-1$

Obtained state of the system (9) will be determined by solving equations:

$$\left\{ \begin{aligned} -b_{11}x_{1s}^3 - b_{11}x_{2s}^3 - b_{11}k_{12}x_{1s}x_{2s} + (a_{11} + b_{11}k_1)x_{1s} + (a_{12} + b_{11}k_2)x_{2s} + a_{13}x_{3s} + \dots + a_{1n}x_{ns} &= 0 \\ -b_{22}x_{1s}^3 - b_{22}x_{2s}^3 - b_{22}k_{12}x_{1s}x_{2s} + (a_{21} + b_{22}k_1)x_{1s} + (a_{22} + b_{22}k_2)x_{2s} + a_{23}x_{3s} + \dots + a_{2n}x_{ns} &= 0 \\ \dots \dots \dots \dots \dots \dots \dots \dots \dots \dots &\dots \\ a_{n1}x_{1s} + a_{n2}x_{2s} + \dots - b_{nn}x_{ns}^3 - b_{nn}x_{n-1s}^3 - b_{nn}k_{n-1,n}x_{n-1s}x_{ns} + (a_{n,n-1} + b_{nn}k_{n-1})x_{n-1s} + (a_{nn} + b_{nn}k_n)x_{ns} &= 0 \end{aligned} \right. \quad (10)$$

From the system of equations (10) we can find the stationary states:

$$x_{1s} = 0, x_{2s} = 0, \dots, x_{ns} = 0, \quad (11)$$

Some other stationary states can be determined by solving equations:

$$-b_{ii}x_{is}^2 + a_{ii} + b_{ii}k_i = 0, x_{js} = 0 \quad \text{at} \quad i \neq j, i = 1, \dots, n \quad (12)$$

Equation (12) with negative $k_i + \frac{a_{ii}}{b_{ii}}$, $i = 1, \dots, n$, $(k_i + \frac{a_{ii}}{b_{ii}} < 0, i = 1, \dots, n)$ have imaginary solutions that cannot correspond to

any physical possible situation. When $k_i - \frac{a_{ii}}{b_{ii}}$ greater than zero ($k_i - \frac{a_{ii}}{b_{ii}} > 0$), equation (12) admits the following steady states:

$$x_{is} = \sqrt{k_i + \frac{a_{ii}}{b_{ii}}}, x_{js} = 0 \quad \text{at} \quad i \neq j, i = 1, \dots, n; j = 1, \dots, n \quad (13)$$

and

$$x_{is} = -\sqrt{k_i + \frac{a_{ii}}{b_{ii}}}, x_{js} = 0 \quad \text{at} \quad i \neq j, i = 1, \dots, n; j = 1, \dots, n \quad (14)$$

$$\begin{cases} \dot{x} = -b_{11}x_1^3 - b_{11}x_2^3 - b_{11}k_{12}x_1x_2 - 3b_{11}\sqrt{k_1 + \frac{a_{11}}{b_{11}}x_1^2} - 3b_{11}\sqrt{k_2 + \frac{a_{12}}{b_{11}}x_2^2} - \\ - 2b_{11}\left(k_1 + \frac{a_{11}}{b_{11}}\right)x_1 - 2b_{11}\left(k_2 + \frac{a_{12}}{b_{11}}\right)x_2 + a_{13}x_3 + \dots + a_{1n}x_n \\ \dot{x} = -b_{22}x_1^3 - b_{22}x_2^3 - b_{22}k_{12}x_1x_2 - 3b_{22}\sqrt{k_1 + \frac{a_{21}}{b_{11}}x_1^2} - 3b_{22}\sqrt{k_2 + \frac{a_{22}}{b_{22}}x_2^2} - \\ - 2b_{22}\left(k_1 + \frac{a_{21}}{b_{22}}\right)x_1 - 2b_{22}\left(k_2 + \frac{a_{22}}{b_{22}}\right)x_2 + a_{23}x_3 + \dots + a_{2n}x_n \\ \dots \dots \dots \dots \dots \\ \dot{x}_n = a_{n1}x_1 + \dots + a_{n2}x_2 + \dots + b_{nn}x_{n-1}^3 - b_{nn}x_n^3 - b_{nn}k_{n-1}x_{n-1}x_n - \\ - 3b_{nn}\sqrt{k_{n-1} + \frac{a_{n,n-1}}{b_{n-1,n}}x_{n-1}^2} - 3b_{nn}\sqrt{k_n + \frac{a_{nn}}{b_{nn}}x_n^2} - 2b_{nn}\left(k_{n-1} + \frac{a_{n,n-1}}{b_{nn}}\right)x_{n-1} - 2b_{nn}\left(k_n + \frac{a_{nn}}{b_{nn}}\right)x_n \end{cases} \quad (20)$$

The control system of deterministic chaotic modes of nonlinear objects with a single input and a single output in the class of hyperbolic umbilic catastrophe.

Find the total time derivative of the Lyapunov vector-function as the dot product of the velocity vector and the gradient vector:

$$\begin{aligned} \frac{dV(x)}{dt} = & -b_{11}^2 \left[x_1^3 + \frac{1}{2}k_{12}x_1x_2 + 3\sqrt{k_1 + \frac{a_{11}}{b_{11}}x_1^2} + 2\left(k_1 + \frac{a_{11}}{b_{11}}\right)x_1 \right]^2 - \\ & -b_{11}^2 \left[x_2^3 + \frac{1}{2}k_{12}x_1x_2 + 3\sqrt{k_2 + \frac{a_{12}}{b_{11}}x_2^2} + 2\left(k_2 + \frac{a_{12}}{b_{11}}\right)x_2 \right]^2 - a_{13}^2x_3^2, \dots, -a_{1n}^2x_n^2 \\ & -b_{22}^2 \left[x_1^3 + \frac{1}{2}k_{12}x_1x_2 + 3\sqrt{k_1 + \frac{a_{21}}{b_{22}}x_1^2} + 2\left(k_1 + \frac{a_{21}}{b_{22}}\right)x_1 \right]^2 - \\ & -b_{22}^2 \left[x_2^3 + \frac{1}{2}k_{12}x_1x_2 + 3\sqrt{k_2 + \frac{a_{22}}{b_{22}}x_2^2} \right]^2 - a_{23}^2x_3^2, \dots, -a_{2n}^2x_n^2 - \\ & \dots \dots \dots \dots \dots \\ & -a_{n1}^2x_1^2 - a_{n2}^2x_2^2, \dots, - \\ & -b_{nn}^2 \left[x_{n-1}^3 + \frac{1}{2}k_{n-1,n}x_{n-1}x_n + 3\sqrt{k_{n-1} + \frac{a_{n,n-1}}{b_{nn}}x_{n-1}^2} + 2\left(k_{n-1} + \frac{a_{n,n-1}}{b_{nn}}\right)x_{n-1} \right]^2 - \\ & -b_{nn}^2 \left[x_n^3 + \frac{1}{2}k_{n-1,n}x_{n-1}x_n + 3\sqrt{k_n + \frac{a_{nn}}{b_{nn}}x_n^2} + 2\left(k_n + \frac{a_{nn}}{b_{nn}}\right)x_n \right]^2, \end{aligned} \quad (21)$$

From (21) it is clear that the total time derivative of the Liapunov vector-function is a negative function, therefore, a sufficient condition is met for asymptotic system stability (20).

Using components of Lyapunov vector-function we construct the Lyapunov vector-function in the scalar form:

$$\begin{aligned} V(x) = & \frac{1}{4}b_{11}x_1^4 + \frac{1}{4}b_{11}k_{12}x_1^2x_2 + b_{11}\sqrt{k_1 + \frac{a_{11}}{b_{11}}x_1^2} + b_{11}\left(k_1 + \frac{a_{11}}{b_{11}}\right)x_1^2 + \\ & + \frac{1}{4}b_{22}x_2^4 + \frac{1}{4}b_{22}k_{12}x_1^2x_2 + b_{22}\sqrt{k_2 + \frac{a_{12}}{b_{22}}x_2^2} + b_{22}\left(k_2 + \frac{a_{12}}{b_{22}}\right)x_2^2 - \frac{1}{2}a_{13}x_3^2 - \dots - a_{1n}x_n^2 + \dots + \\ V(x) = & -\frac{1}{2}a_{n1}x_1^2 - \frac{1}{2}a_{n2}x_2^2 - \dots + \frac{1}{4}b_{nn}x_{n-1}^4 + \frac{1}{4}b_{nn}k_{n-1}x_{n-1}^2x_n + b_{nn}\sqrt{k_{n-1} + \frac{a_{n,n-1}}{b_{nn}}x_{n-1}^2} + \\ & + b_{nn}\left(k_{n-1} + \frac{a_{n,n-1}}{b_{nn}}\right)x_{n-1}^2 + \frac{1}{4}b_{nn}x_n^4 + \frac{1}{4}b_{nn}k_{n-1,n}x_{n-1}^2x_n + b_{nn}\sqrt{k_n + \frac{a_{nn}}{b_{nn}}x_n^2} + b_{nn}\left(k_n + \frac{a_{nn}}{b_{nn}}\right)x_n^2 \end{aligned} \quad (22)$$

From the expressions (22) positive or negative definition of Lyapunov function cannot be defined, so we use the

fundamental theorem of the theory of catastrophes - Morse Lemma [19-21]. By Lemma Morse, Lyapunov function (22) in the vicinity of the stationary state (13) and (14) can be locally presented in the quadratic form considering the state equation (20) to deviations with respect to the stationary state (13) or (14):

$$\begin{aligned} V(x) = & \left[(b_{11} + b_{22})k_1 + a_{11} + a_{21} - \frac{1}{2}(a_{31} + a_{41} + \dots + a_n) \right] x_1^2 + \\ & + \left[(b_{11} + b_{22})k_2 + a_{12} + a_{22} - \frac{1}{2}(a_{32} + a_{42} + \dots + a_{n2}) \right] x_2^2 + \\ & + \left[(b_{33} + b_{44})k_3 + a_{33} + a_{43} - \frac{1}{2}(a_{13} + a_{23} + \dots + a_{n3}) \right] x_3^2 + \dots + \\ & + \left[(b_{33} + b_{44})k_4 + a_{44} + a_{34} - \frac{1}{2}(a_{14} + a_{24} + \dots + a_{n4}) \right] x_4^2 + \dots + \\ & + \left[(b_{n-1,n-1} + b_{nn})k_{n-1} + a_{n-1,n-1} + a_{n,n-1} - \frac{1}{2}(a_{1,n-1} + a_{2,n-1} + a_{3,n-1} + \dots + a_{n,n-1}) \right] x_{n-1}^2 + \\ & + \left[(b_{n-1,n-1} + b_{nn})k_n + a_{n-1,n} + a_{nn} - \frac{1}{2}(a_{1n} + a_{2n} + a_{3n} + \dots + a_{n-2,n}) \right] x_n^2, \end{aligned} \quad (23)$$

Stability conditions of stationary state (13) or (14) are determined by the positive definiteness of the quadratic form (23), i.e. by the system of inequalities:

$$\begin{cases} (b_{11} + b_{22})k_1 + a_{11} + a_{21} > \frac{1}{2}(a_{31} + a_{41} + \dots + a_{n1}) \\ (b_{11} + b_{22})k_2 + a_{12} + a_{22} > \frac{1}{2}(a_{32} + a_{42} + \dots + a_{n2}) \\ (b_{33} + b_{44})k_3 + a_{33} + a_{43} > \frac{1}{2}(a_{13} + a_{23} + \dots + a_{n3}) \\ (b_{33} + b_{44})k_4 + a_{34} + a_{44} > \frac{1}{2}(a_{14} + a_{24} + \dots + a_{n4}) \\ \dots \dots \dots \dots \dots \\ (b_{n-1,n-1} + b_{nn})k_{n-1} + a_{n-1,n-1} + a_{n,n-1} > \frac{1}{2}(a_{1,n-1} + a_{2,n-1} + \dots + a_{n,n-1}) \\ (b_{n-1,n-1} + b_{nn})k_n + a_{n-1,n} + a_{nn} > \frac{1}{2}(a_{1n} + a_{2n} + \dots + a_{n-2,n}) \end{cases} \quad (24)$$

Stability conditions of stationary state (13) or (14) are determined by the positive definiteness of the quadratic form (23), i.e. by the system of inequalities (24).

Thus, nonlinear MIMO system constructed in the class of three-parameter structurally stable maps will be stable in the unlimited wide range of changes in the undetermined parameters of the control object. The steady state (11) exists and it is stable when changing the undetermined parameters in the region (20), whereas stationary state (13) and (14) exists in case of the loss of stability in the stationary state (11) and they do not exist simultaneously.

Stable stationary states (13) and (14) can be obtained when inequalities (24) are true.

III. CONCLUSIONS

The real control objects are nonlinear, multidimensional and their control systems are designed and operated under conditions of uncertainty. The basic property of nonlinear dynamic control objects is functioning in the deterministic chaos mode with attainment of "strange attractor". The deterministic chaotic modes of control objects can lead to accidents and crisis, and the chaos is shown in the form of loss of stability of the existing stationary state of the system in the conditions of uncertainty.

We propose to solve control problems of deterministic chaotic processes by building control systems with a high potential for robust stability in the class of hyperbolic umbilic catastrophe. The study of control system is performed using gradient-velocity method of the Vector Lyapunov function.

The stability region of stationary state in the control system is obtained in the form of a system of inequalities for the simplest uncertain parameters of the closed system. The control system with a high potential for robust stability, built in the class of hyperbolic umbilic catastrophe provides robust stability for any changes in uncertain parameters. Hence, a deterministic chaotic mode is removed from the scenario of development process.

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